

# The Welding and Solidification Metallurgy of Alloy 625

*Chemical composition and solidification microstructure are correlated to hot cracking susceptibility*

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**ABSTRACT.** The weld metal microstructure development and solidification cracking behavior of Alloy 625 gas tungsten arc (GTA) welds as a function of composition has been determined. A three-factor, two-level, factorially-designed set of alloys involving the elements C, Si and Nb was examined. Differential thermal analysis (DTA) of these alloys indicated that Nb, and to a lesser extent C and Si, increased the melting/solidification temperature range. The DTA revealed that terminal solidification constituents were formed in the Nb-bearing alloys, the presence of which was confirmed by optical and electron microscopy techniques and identified as  $\gamma$ /MC-(NbC) carbide,  $\gamma$ /Laves and  $\gamma$ /M<sub>6</sub>C carbide eutectic-type constituents. Addition of carbon to the Nb-bearing alloys was observed to promote the formation of the  $\gamma$ /MC(NbC) carbide constituent and Si was observed to promote increased formation of the  $\gamma$ /Laves constituent. Regression analysis of Varestraint hot-crack testing data revealed that additions of C or Si to Alloy 625 increased the susceptibility of the alloy to hot cracking. Niobium-free alloys were observed to have a very low tendency toward solidification hot cracking, but even among these alloys, C and Si additions were detrimental. It was concluded that the increased solidification temperature range and formation of Nb-rich eutectic constituents were primarily responsible for the increased susceptibility of Nb-bearing alloys to solidification cracking.

## Introduction

Alloy 625 (58 minimum Ni-20-23 Cr, 8-10 Mo, 3.15-4.15 Nb+Ta-5 maximum Fe-0.5 maximum Mn, 0.5 maximum Si, 0.10 maximum C wt-%) has been a commonly used nickel-based alloy for over two de-

acades. Although originally developed as a turbine alloy (Ref. 1), its combination of good oxidation and corrosion resistance and moderate mechanical strength have made it a successful alloy in many other applications. Among these are cladding and surfacing for marine environments (Refs. 2, 3) and for wear resistance as hardfacing for tool and die steels (Ref. 4).

Alloy 625 is not without its problems, though. Recent studies (Refs. 5, 6), have indicated that this alloy can be susceptible to hot cracking. Patterson and Milewski (Ref. 5) noted that hot cracked surfaces in arc welds made between Alloy 625 and 304L stainless steel were enriched in S, Nb, P and C, and that eutectic-like structures were present in the microstructures of these welds. Cieslak, *et al.* (Ref. 7), found that dissimilar metal CO<sub>2</sub> laser beam welds between Inconel Alloy 625 and 304 stainless steel made at slow travel speed (10 in./min) contained a Nb-rich Laves phase. Although in comparison to many other nickel alloys (Refs. 8, 9), Alloy 625 has a good reputation for resistance to hot cracking, it appears from the literature not to be totally immune from the problem.

A review of the literature reveals no published study that correlates the solidification microstructure with weldability in Alloy 625 as a function of chemical composition. In addition, no published report describes the sequences of solidification events leading to the development of the

observed microstructure in Alloy 625. The purpose of this work, then, is twofold. First, the welding metallurgy of Alloy 625 is described in some detail. That is, the evolution of weld metal microstructure upon cooling from the liquidus is explained. Second, a correlation is established between alloy chemistry and both solidification microstructure and weldability (hot-cracking susceptibility). These results may provide for intelligent alloy optimization schemes for Alloy 625 and similar materials from a weldability perspective.

## Experimental Procedure

The alloy design decision for this experiment was driven by the desire to establish the fundamental solidification and hot-cracking mechanisms in this alloy system. The effect of tramp elements (S, P, B, etc.) on the solidification and weldability behavior of nickel-based alloys has been well established. It was concluded that studying these elements would not add much new insight. They were eliminated from this study by being held at constant low levels. Also, it had been observed earlier (Ref. 10) that the solidification microstructure in Alloy 625 and similar alloys could contain minor constituents composed of Laves phase and MC carbide. Both of these phases were Nb-rich and Laves had been shown (Ref. 11) to be stabilized by Si alloying additions. Based upon these two factors, C, Si and Nb were chosen as the composition variables for this experiment.

A three-factor, two-level factorial series of alloys was designed around these elements. The aim low level for C, 0.005 wt-%, was effectively the limit of industrial processing capabilities. The high level for C was set at what would likely be expected in a high C commercial heat of Alloy 625, 0.035 wt-%. The low level for Si was set as none intentionally added. The high level aim was 0.35 wt-%. The low level for Nb was set as none intentionally added. The high level aim was 3.5 wt-%. The levels for Nb would clearly indicate the difference between Nb-bearing and

## KEY WORDS

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Solidification Crack  
GTA Alloy 625 Welds  
Differential Analysis  
Thermal Analysis  
Eutectic Constituent  
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 $\gamma$ /M<sub>6</sub> Carbide  
Nb-Containing Alloys

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Table 1—Alloy Compositions (wt-%)

| Element | Alloy Number |       |       |       |       |       |       |       |
|---------|--------------|-------|-------|-------|-------|-------|-------|-------|
|         | 1            | 2     | 3     | 4     | 5     | 6     | 7     | 8     |
| C       | 0.006        | 0.031 | 0.006 | 0.036 | 0.009 | 0.038 | 0.008 | 0.035 |
| Mn      | 0.02         | 0.02  | 0.02  | 0.03  | 0.03  | 0.03  | 0.03  | 0.03  |
| P       | 0.005        | 0.005 | 0.005 | 0.005 | 0.006 | 0.006 | 0.006 | 0.006 |
| S       | 0.002        | 0.002 | 0.002 | 0.003 | 0.003 | 0.003 | 0.004 | 0.003 |
| Si      | 0.03         | 0.03  | 0.35  | 0.39  | 0.03  | 0.03  | 0.38  | 0.46  |
| Cr      | 22.10        | 21.95 | 21.63 | 21.57 | 21.81 | 21.83 | 21.65 | 21.68 |
| Mo      | 9.54         | 9.61  | 9.60  | 9.63  | 9.81  | 9.81  | 9.68  | 9.67  |
| Ti      | 0.06         | 0.06  | 0.06  | 0.06  | 0.06  | 0.06  | 0.06  | 0.06  |
| Nb      | 0.01         | 0.01  | 0.02  | 0.02  | 3.61  | 3.60  | 3.57  | 3.53  |
| Fe      | 2.56         | 2.55  | 2.18  | 2.59  | 2.30  | 2.31  | 2.26  | 2.29  |
| Ni      | bal          | bal   | bal   | bal   | bal   | bal   | bal   | bal   |

Nb-free alloys. The alloys were double vacuum melted at the Sandia National Laboratories melting and solidification facility. Initial melting was done in a vacuum induction furnace from virgin raw materials. Electrodes weighing approximately 150 lb (68.2 kg) were poured in vacuum. These electrodes were then vacuum arc remelted to 6-in. (152-mm) diameter ingots in preparation for hot working. Table 1 lists the compositions of the eight alloys studied.

Hot working of the ingots began with extrusion at 1175°C (2147°F) down to a 3-in. (76-mm) diameter bar. These bars were then flattened at 1175°C to approximately 0.6-in. plates. From these plates, specimens were taken for differential thermal analysis (DTA). The plates were further reduced by hot rolling (1175°C) to a thickness of approximately 0.18 in. (4.6 mm). Further reduction was done at room temperature to a thickness of approxi-

mately 0.12 in. (3 mm). These sheets were given a final anneal at 1010°C (1850°F) and water quenched. A cold straightening pass (<1% cold work) was then made to prepare the sheets for machining into Vareststraint test specimens.

The autogenous gas tungsten arc (GTA) Vareststraint test was used to quantify the susceptibility of these alloys to fusion zone hot cracking. The GTA welding parameters used were 100 A, (direct current, electrode negative at a travel speed of 8 in./min. The machine voltage was ≈ 12 V and argon was used as the shielding gas. These conditions produced welds that were approximately 0.20 in. wide at the top surface. The test specimens measured ≈ 6.5 X ≈ 1 X ≈ 0.12 in. (165 X 25 X 3 mm). All tests were performed at a strain level of ≈ 2.5% to simulate high restraint welding conditions. Replicate testing (4 to 5 tests per alloy) was employed to develop acceptable statistics. The order of testing was randomized to eliminate systematic error. Maximum crack length (MCL) was the quantitative measure of hot-cracking susceptibility used in this study.

Differential thermal analysis (DTA) test-

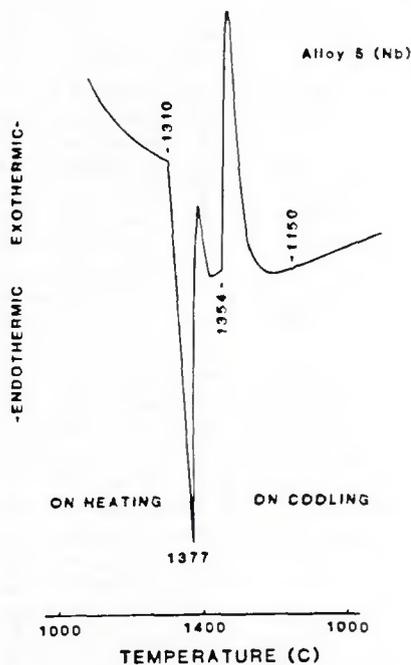


Fig. 1—DTA thermogram (20° C/min) for Alloy 5.

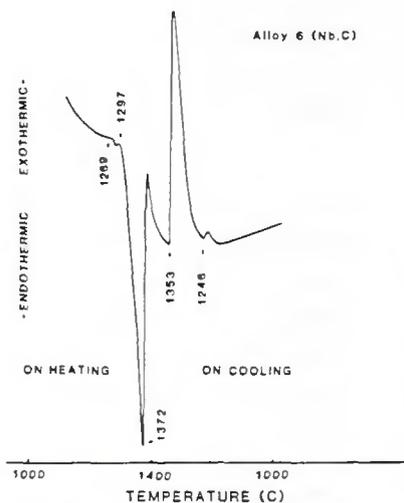


Fig. 2—DTA thermogram (20° C/min) for Alloy 6.

ing was done on a Netsch thermal analyzer STA 429. Samples were machined from blocks taken from the hot-worked plates that had been subsequently annealed in vacuum at 1200°C (2192°F) for 4 h and water quenched. The samples weighed ≈ 0.8 g. All tests were conducted in a helium environment with pure W used as the reference material. To calibrate the system, pure Ni was found to melt within 2°C (3.6°F) of the established literature value. The experiments involved heating and cooling the Nb-bearing alloys (Alloys 5-8) through the melting/solidification temperature range as fast as was possible (20°C/min - 36°F/min) with the available equipment. The purpose of these experiments was to identify any terminal solidification reactions that were occurring in these alloys.

Vareststraint test and DTA specimens were examined metallographically. After polishing through 0.05 μm alumina, they were electroetched (1-2 V) with 10% chromic acid for times necessary to reveal the structures of interest. Carbon-coated metallographic specimens were examined in the Hitachi 500 scanning electron microscope (SEM). Areas of interest were also analyzed with a Cameca MBX electron microprobe. In all cases, specimens were repolished flat and carbon coated prior to microprobe analysis. Point count analyses and profiles were carried out under the operating conditions of 15 kV and a beam current of ≈ 20 nA. K<sub>α</sub> x-ray lines were used for analysis of all elements except Mo and Nb where L<sub>α</sub> x-ray lines were used. The raw counting data were converted to weight percentages with a φ(ρ,Z) correction algorithm (Ref. 12).

High-resolution analyses of weld metal microstructures were performed using transmission electron microscopy (TEM) and analytical electron microscopy (AEM). Both thin foils and extraction replicas were examined. Specimens were examined in both a JEOL 200CX operating at 200 kV and a JEOL 100CX operating at 120 kV and equipped with a Tracor Northern EDS detector/spectrometer. Selected-area electron diffraction was used to identify the crystal structures of the various phases examined.

X-ray spectra collected in the AEM were reduced to weight percentages with the Cliff-Lorimer standardless-ratio technique (Ref. 13) with a data reduction program developed by Romig (Ref. 14). The Cliff-Lorimer k-factors were obtained experimentally for all elements analyzed except Si, where a suitable standard material was not available. In the case of Si, a calculated (Ref. 14) k-factor was used. K<sub>α</sub> x-ray lines were used for the analysis of all elements. An absorption correction was not needed as the thin foil condition was never violated (Ref. 15). The AEM compositional data were obtained for the elements Ni, Fe, Cr, Nb, Mo and Si.





tions consistent with the two minor peaks on the DTA thermogram—Fig. 4. Again, the Laves is found to be enriched in Si—Table 3. With Si present in the alloy, the  $\gamma$ /Laves constituent is not precluded from forming even though C is also intentionally alloyed. A more detailed description of the evolution of the solidification microstructure in these alloys has been given recently by Cieslak, *et al.* (Ref. 16).

The microstructural modification in these alloys as a function of the minor elements C and Si is remarkable but not without precedence. Nakao, *et al.* (Ref. 20), studied the effects on microstructure caused by alloying IN-519 (Fe-24Cr-24Ni-3Nb wt-%) with C and Si. At low C levels (0.02-0.03 wt-%),  $\gamma$ /Laves was the only solidification constituent observed in weld metal. As C content was increased (>0.2 wt-%) the  $\gamma$ /Laves constituent was replaced by  $\gamma$ /MC (NbC) constituent in a manner analogous to that observed in the present study. The  $\gamma$ /MC constituent solidified at a higher temperature than the  $\gamma$ /Laves constituent, also consistent with the results of the present work on Alloy 625.

In summary, the evolution of solidification microstructure in Alloy 625 fusion welds is dominated by the segregation of Nb. The particular interdendritic constituents observed (e.g.,  $\gamma$ /Laves and  $\gamma$ /MC) in a given heat of Alloy 625 are influenced by the concentration of the elements C and Si. These constituents are not observed when Nb is not present.

#### Weldability Analysis

In the early development of analysis of Varestraint test data, Savage and Lundin (Ref. 21) interpreted maximum crack length as a measure of the hot-cracking temperature range. More recently, a similar approach has been adopted by Matsuda and coworkers (Ref. 22) in their interpretation of trans-Varestraint test data. Their brittleness temperature range (BTR) parameter is obtained from the maximum crack length achieved at high levels of applied strain during trans-Varestraint testing. From a metallurgical viewpoint, they suggested that the BTR could be correlated to the existence of a two-phase, liquid-plus-solid structure in an alloy. The lower limit of the BTR could extend down to the terminal solidification reaction of a nonequilibrium eutectic-type constituent.

The general correlation between weldability and solidification behavior can be traced back to early investigators (Refs. 23, 24) who concluded that alloys having wider solidification temperature ranges would be more crack susceptible than alloys having narrow solidification temperature ranges. The first to quantify this understanding was Borland (Ref. 24), who attempted to determine the hot-cracking

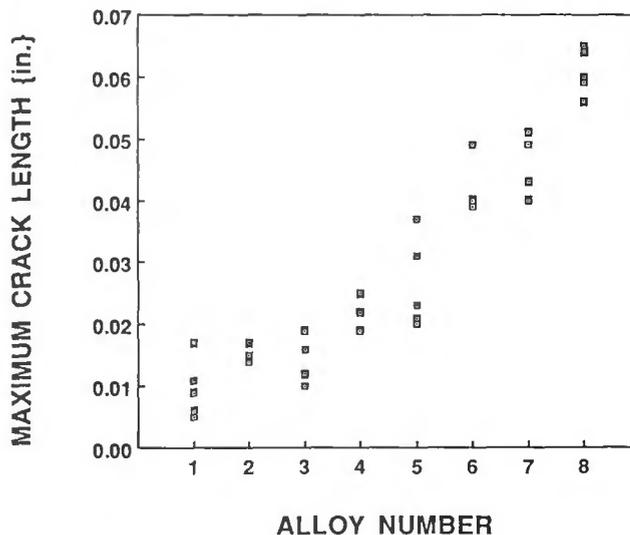


Fig. 6—Varestraint test results for Alloys 1-8.

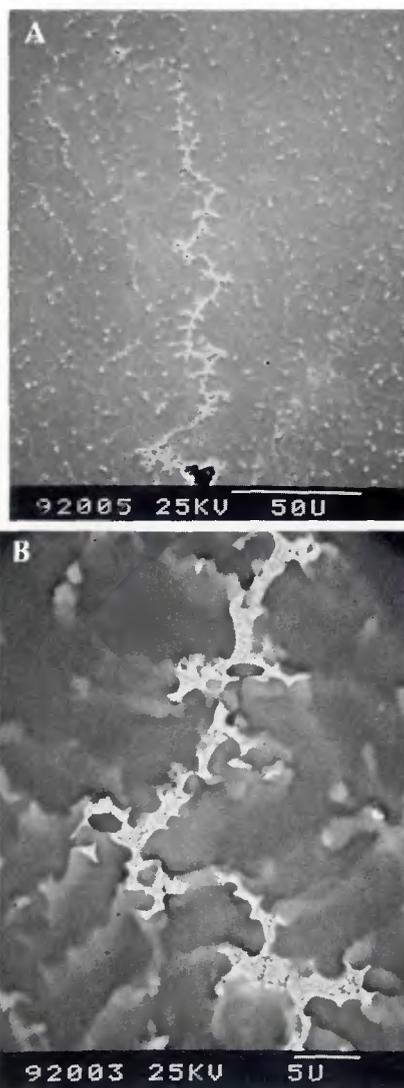


Fig. 7—A—SEM image of microstructure associated with hot cracks in Alloy 7; B—higher magnification SEM image showing constituent morphology in greater detail.

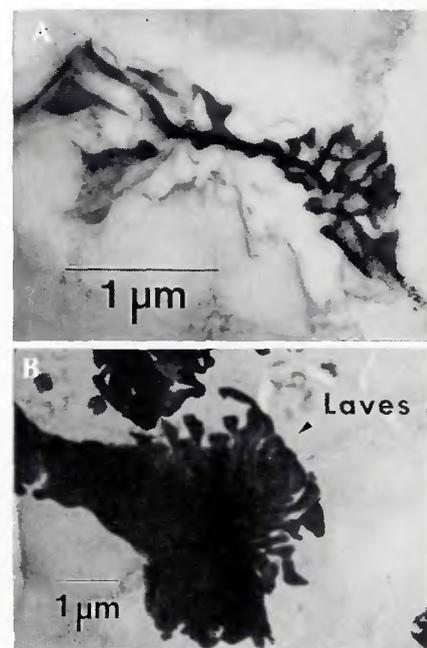
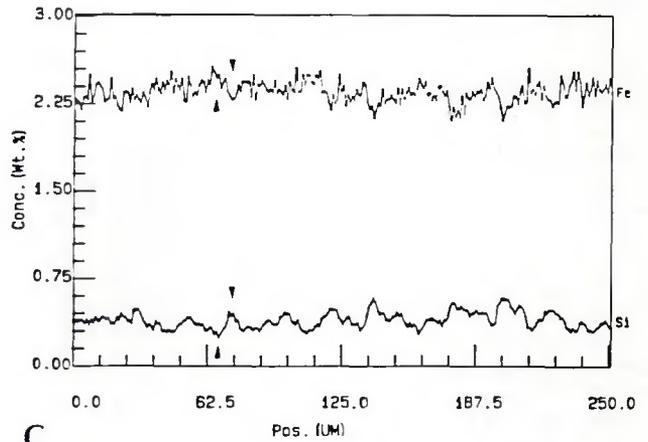
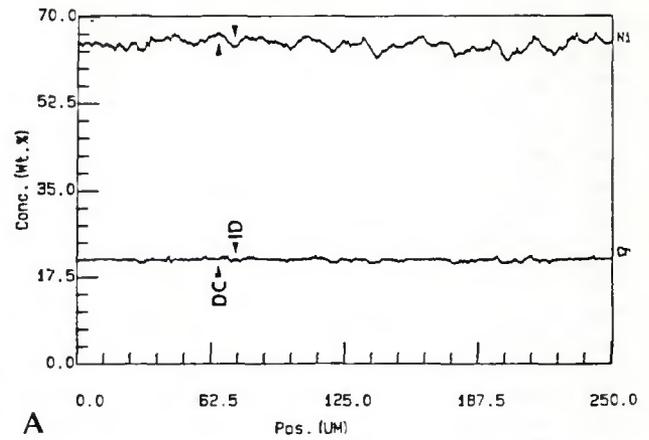
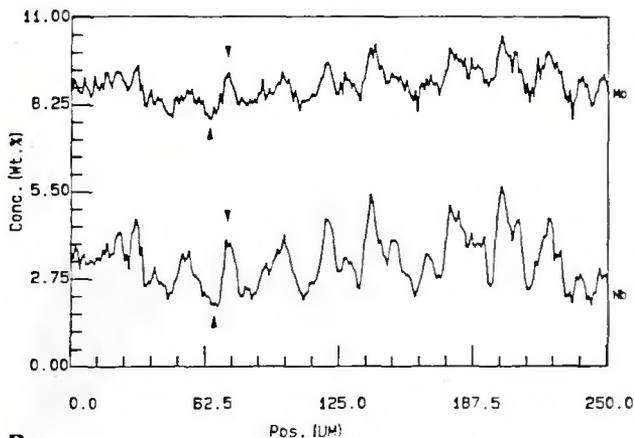


Fig. 8—A—TEM micrograph of MC carbide morphology in Alloy 6 Varestraint test specimen; B—TEM micrograph of Laves morphology in Alloy 7 Varestraint test specimen.

Fig. 9—Microprobe profile showing dendritic segregation pattern in Alloy 7. A—Ni, Cr; B—Mo, Nb; C—Fe, Si. DC indicates a dendrite core position, and ID an interdendritic position.



susceptibility of binary alloy systems from their equilibrium phase diagrams. Alloys that had extensive temperature separation between their liquidus and solidus were likely to be hot-crack sensitive. Using available binary phase diagrams, Borland developed a quantitative descriptor of the effect of a given alloying element on the equilibrium melting/solidification temperature range. He called this parameter the relative potency factor (Ref. 24) (RPF). It is defined as follows:

$$RPF = m_L (1-k)/k \quad (1)$$

where  $m_L$  is the liquidus slope and  $k$  is the

equilibrium distribution coefficient. The units of the RPF are  $^{\circ}\text{C}/\text{wt}\%$ . When the RPF is multiplied by the nominal alloy concentration, the equilibrium melting/solidification temperature range,  $\Delta T_{eq}$ , results.

For the present alloys, the equilibrium melting temperature ranges,  $\Delta T_{eq}$ , have been determined previously by Cieslak, *et al.* (Ref. 16). These data are listed in Table 5 and reveal that the Nb-bearing alloys (5-8) have a larger melting temperature range than the Nb-free alloys (1-4). (A melting temperature range for commercial Alloy 625 has been reported as  $\approx 60^{\circ}\text{C}/108^{\circ}\text{F}$ —Ref. 25.) In particular, Alloy 1, the base alloy of the experiment, has an average  $\Delta T_{eq}$  of only  $26.5^{\circ}\text{C}$  ( $48^{\circ}\text{F}$ ), in comparison to Alloy 5 (which contains the high level of Nb), which has a  $\Delta T_{eq}$  of  $55.3^{\circ}\text{C}$  ( $100^{\circ}\text{F}$ ). A functional relationship between  $\Delta T_{eq}$  and alloy composition was obtained (Ref. 16) by performing regression analysis on the data in Table 5, and is given below.

$$\Delta T_{eq} (^{\circ}\text{C}) = 20.9 + 405.5 (\text{wt}\% \text{C}) + 20.3 (\text{wt}\% \text{Si}) + 9.4 (\text{wt}\% \text{Nb}) \quad (2)$$

The coefficient of correlation ( $R^2$ ) for this relationship is 0.98 and the standard error is  $3.0^{\circ}\text{C}$  ( $5.4^{\circ}\text{F}$ ).

From these observations and perspec-

tives, then, the hot-cracking response of the eight alloys examined here can begin to be understood. It is clear from examining Fig. 6 that the Nb-bearing alloys (5-8) are more susceptible to fusion zone hot cracking than are the Nb-free alloys (1-4). It is also clear that the least crack-sensitive alloy is Alloy 1, which has the lowest concentration of C, Si and Nb of any of the alloys examined. The results of regression analysis, given below, performed on the data shown in Fig. 6 reveal in more detail the functional relationship between the variable alloying elements (C, Si, Nb) and the MCL obtained during Vareststraint testing.

$$\text{MCL (in.)} = 0.001 + 0.347 (\text{wt}\% \text{C}) + 0.037 (\text{wt}\% \text{Si}) + 0.007 (\text{wt}\% \text{Nb}) \quad (3)$$

The coefficient of correlation ( $R^2$ ) for this relationship is 0.89 and the standard error is 0.006 in. (0.15 mm). A more complete description of the cracking behavior results when the interactive terms are included in the regression equation, as given below.

$$\text{MCL (in.)} = 0.008 + 0.235 (\text{wt}\% \text{C}) + 0.013 (\text{wt}\% \text{Si}) + 0.004 (\text{wt}\% \text{Nb}) + 0.068 (\text{wt}\% \text{Nb}) (\text{wt}\% \text{C}) + 0.011 (\text{wt}\% \text{Nb}) (\text{wt}\% \text{Si}) \quad (4)$$

Table 5—Equilibrium Melting Temperature Range

| Alloy No.     | $\Delta T_{eq}$           |
|---------------|---------------------------|
| 1             | 26.5 (0.6) <sup>(a)</sup> |
| 2 (C)         | 33.7 (1.5)                |
| 3 (Si)        | 29.0 (1.6)                |
| 4 (C, Si)     | 43.3 (1.3)                |
| 5 (Nb)        | 55.3 (2.4)                |
| 6 (Nb, C)     | 72.8 (3.2)                |
| 7 (Nb, Si)    | 68.3 (2.5)                |
| 8 (Nb, C, Si) | 76.3 (0.6)                |

(a) Values are averages of multiple tests; values in parentheses are standard deviations; all values in  $^{\circ}\text{C}$ .



